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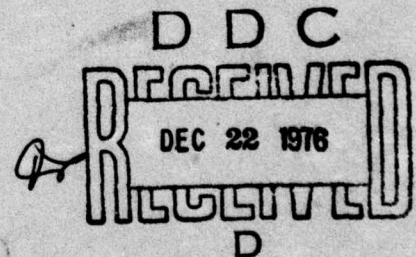


TRANSIENT SUPPRESSION MODIFICATION  
for  
INSTRUMENT LANDING SYSTEM AN/GRN-27(v)  
(Solid State Instrument Landing System (SSILS))

Chester W. Paprocki

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ROME AIR DEVELOPMENT CENTER  
AIR FORCE SYSTEMS COMMAND  
GRIFFISS AIR FORCE BASE, NEW YORK 13441



## PREFACE

The material for this report was collected and prepared under Job Order No. 404L1401. The effort was conducted at the request of the Electronics Systems Division (ESD) TRACALS System Project Office (SPO) in response to direction of the Air Staff to "in-house study, design and test equipment which would reduce the lightning-induced outages of the Solid State Instrument Landing System (SSILS) AN/GRN-27(v)."

The author wishes to acknowledge the assistance and support received from Captain Rudolf Konegen, SPO Program Manager, Joseph J. Polniaszek, Jr., and all the many other people who helped in performing this task.

This report has been reviewed by the Office of Information, RADC, and approved for release to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nations.

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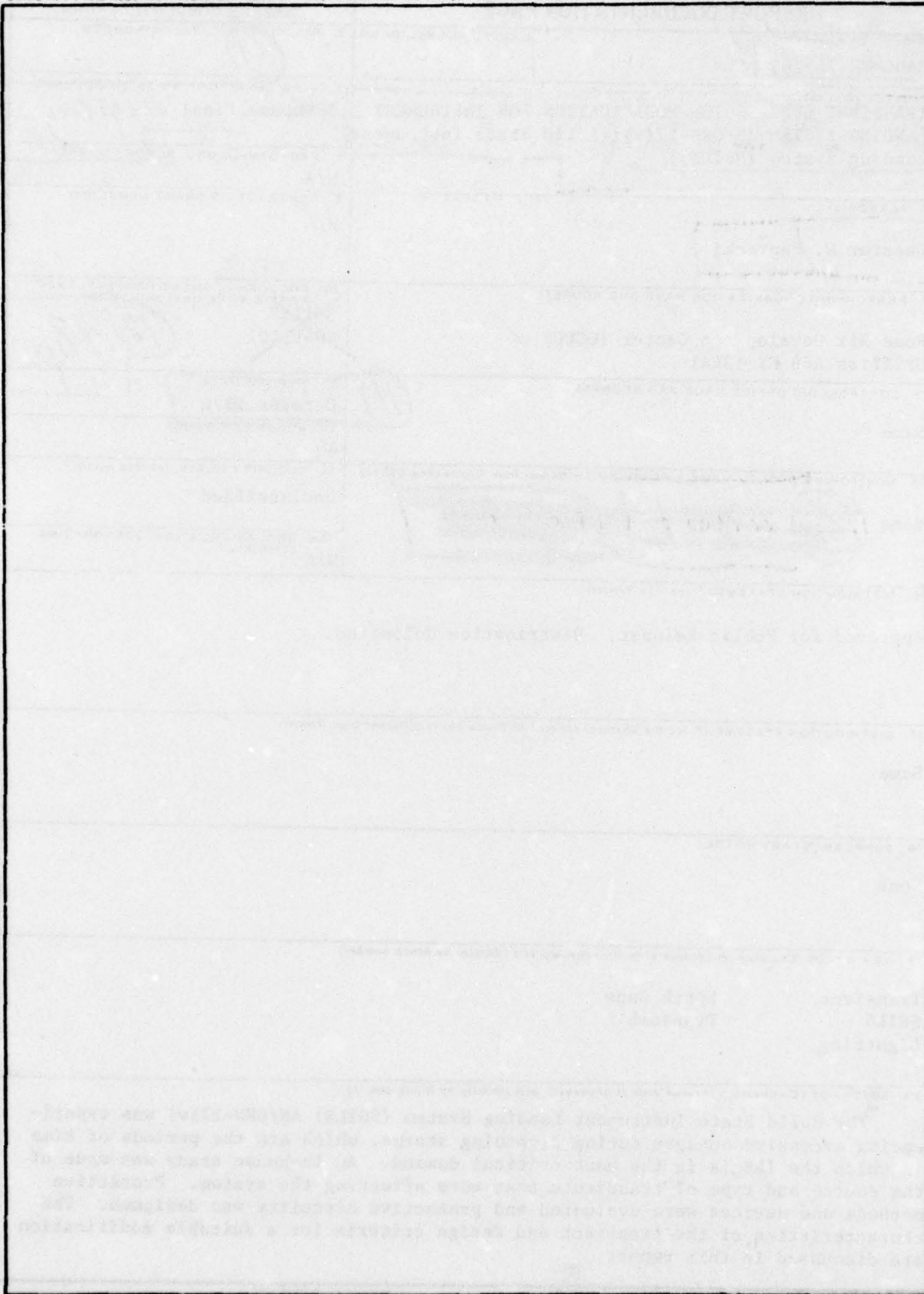
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## I. INTRODUCTION:

The Instrument Landing System AN/GRN-27(v) was designed for worldwide use during Category II weather conditions at intermediate and large airports. The major subsystems of the landing system are dispersed around and often off the airport proper and are interconnected by a network of commercial grade telephone lines. These telephone lines frequently are miles long and routed through various telephone equipment and switching terminals. The interconnecting telephone wires, power wires and antenna arrays of the system all act as transmission lines for induced transients, which may have an adverse effect on the electronics of the landing system.

Shortly after the AN/GRN-27, commonly referred to as the SSILS (Solid State Instrument Landing System), was being commissioned (summer 1973) a high number of disruptions and operational failures were experienced. These failures often occurred during periods of electrical atmospheric disturbances, which were the periods when the ILS was most needed. A study program was initiated to find the reason for the failures and to provide a method of protection. Since there was no evidence of direct lightning hits, it was determined that transients due to electromagnetic disturbances, a result of nearby lightning activity, were exceeding the voltage rating of interface parts of the SSILS.

An analysis was made to determine the severity of the lightning transient problem. Since the behavior of lightning is unpredictable, a set of transient characteristics was developed. The available protection devices were then evaluated to provide circuit protection to isolate the sensitive ILS components from the destructive transients. A multi-stage protection circuit was designed using spark gap type devices and zener diodes. These protection circuits were



breadboard constructed to allow evaluation.

Tests on a hot mock-up of the SSILS indicated that changes could be made to the protection circuit to improve the filter action and enhance the life of the spark gaps. The modified protection circuits were again tested in an operational environment. These tests were conclusive enough to serve as a base for retrofit of the SSILS.

## II. DETAIL STUDY OF THE LIGHTNING PROBLEM:

### Type of Protection Needed

Evidence of SSILS failures was first observed on two systems: one installed in New Jersey and another in Delaware. A view of an isoceraunic map (Figure 1) of the United States shows the incidence of lightning in that area to be comparatively low when related to other parts of the CONUS. The worst case condition exists in the Tampa, Florida area with 100 thunderstorm days per year. The monthly distribution obtained from the Worldwide Airfield Climatic Data Manuals is presented in Table I for MacDill, McGuire and Dover AFB's. An empirical factor of up to 0.5 is used to convert to the number of strokes per square mile at level ground. Using the figures in Table I, it may be determined that 42 strokes/mile<sup>2</sup>/year can be predicted for the MacDill AFB area. In comparison, approximately 15 strokes/mile<sup>2</sup>/year may be expected at McGuire and Dover AFB.

The behavior of lightning is unpredictable. However, the United States Weather Bureau has accumulated statistical data from which one is able to predict the frequency of occurrence of lightning. Bell Laboratories personnel have also calculated, as shown in Figure 2, a distribution of peak initial stroke currents for lightning flashes to buried cables (Reference 2). The numbers presented in the figure give the reader a feel for the magnitude of

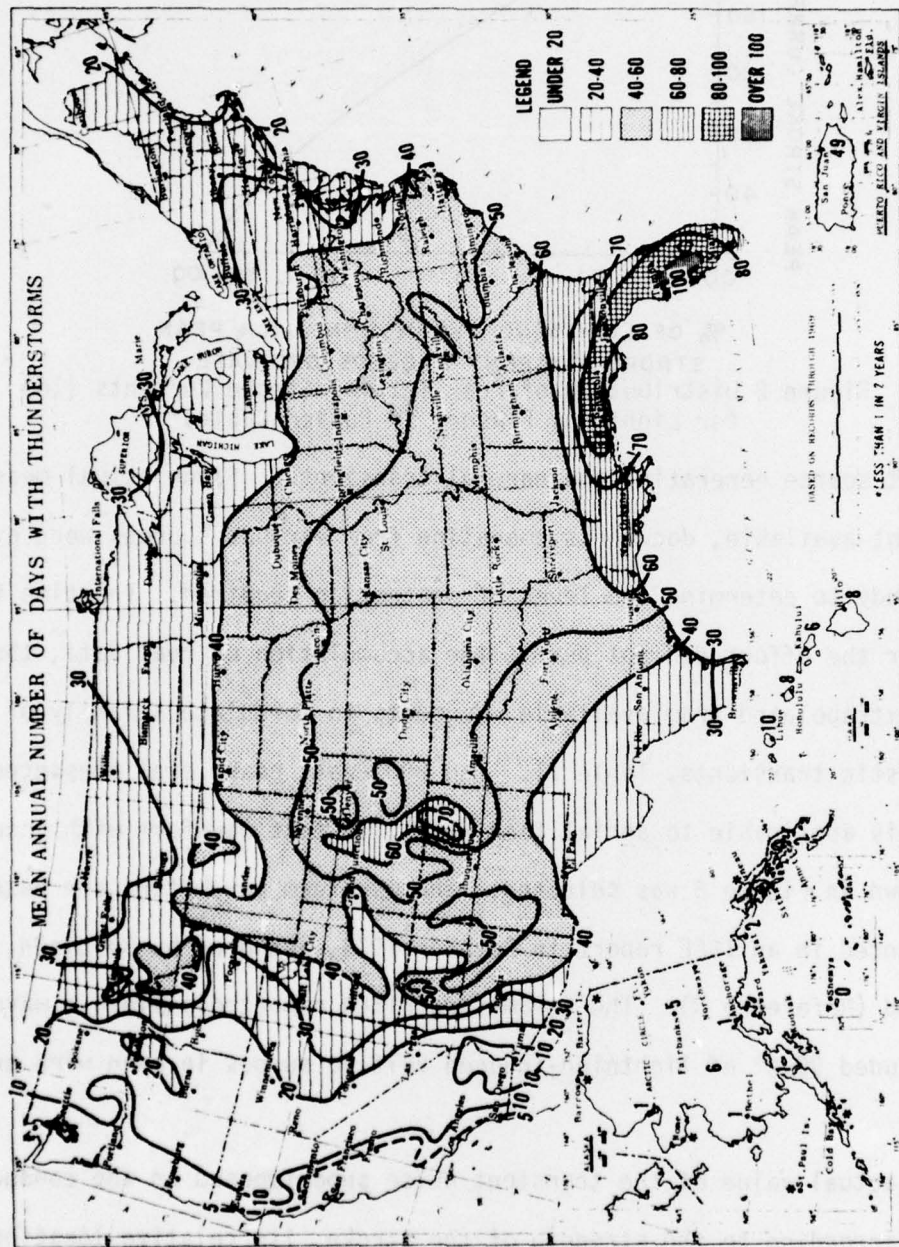
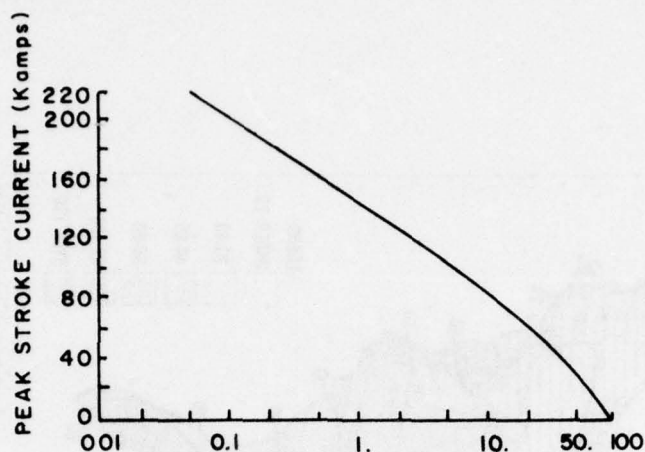


Figure 1 Isoceraunic Map of United States



**% OF LIGHTNING FLASHES IN WHICH PEAK  
STROKE CURRENT EXCEEDS ORDINATE**  
 Figure 2 Distribution of Peak Initial Stroke Currents (10)  
 for Lightning Flashes to Buried Cables\*

the current source generating the harmful transients. Since actual measured data was not available, documents emanating from various sources were evaluated in this study to determine the level of protection required. The time frame allowed for the effort did not permit the accumulation of real data, therefore, data was extrapolated from available documents to formulate a family of characteristic transients, Table II. The 2000-volt peak level presented in the table is applicable to aerial feed wires. A test waveform with characteristics shown in Figure 3 was selected. The waveform parameters are based on data presented in an IEEE report in October 1973, authored by Bennison, Ghazi and Ferland (Reference 3). The parameters cited reportedly yield a waveform which included 99.8% of lightning-induced voltage surges in open wire and cable.

The actual value of the transient pulse superimposed on the conductors will vary according to the strength of the stroke, its relative location to a conductor, the conductivity of surrounding soil and low impedance path to

\*From Wire Journal, Jan. 1973, "Lightning Induced Current Surges on Buried Multicoaxial Cable System," Douglass, D. A. Bell Laboratories



TABLE I

## MEAN NUMBER OF THUNDERSTORM DAYS

	<u>MACDILL AFB</u>	<u>MCGUIRE AFB</u>	<u>DOVER AFB</u>
J	1.1	.1	.2
F	1.9	.4	.3
M	2.6	.7	1.2
A	4.2	2.3	2.6
M	5.9	4.6	5.7
J	13.1	5.5	5.6
J	19.3	6.7	6.6
A	18.2	4.7	5.7
S	12.1	2.3	2.3
O	2.8	1.1	1.1
N	1.1	.2	.4
D	.7	0.0	.1
ANNUAL	83.0	28.6	31.8

Reference -

Worldwide Airfield Climatic Data, USAF Environmental Technical Application Center, Vol. VIII PARTS 6 & 7

TABLE II

## TRANSIENT ENERGY LEVELS

<u>Peak Voltage</u>	<u>Pulse Width</u>	<u>Rise Time</u>	<u>Decay Time</u>	<u>Approx. Energy Level</u>
2,000	250 $\mu$ sec	4 $\mu$ sec	200 $\mu$ sec	33 Joules
1,000	1,000 $\mu$ sec	10 $\mu$ sec	1,000 $\mu$ sec	33 Joules
400	4,000 $\mu$ sec			25 Joules
1,000	100 $\mu$ sec at 100 Hz rate			3.3 Joules

true ground. In the case of the SSILS, the actual magnitudes of the damaging transients were not measured. The values presented in Table II exceed the breakdown voltage of the interface components of the SSILS and encompass the characteristics of known lightning-induced transient pulses, therefore, it was adopted as a guide for purposes of this study.

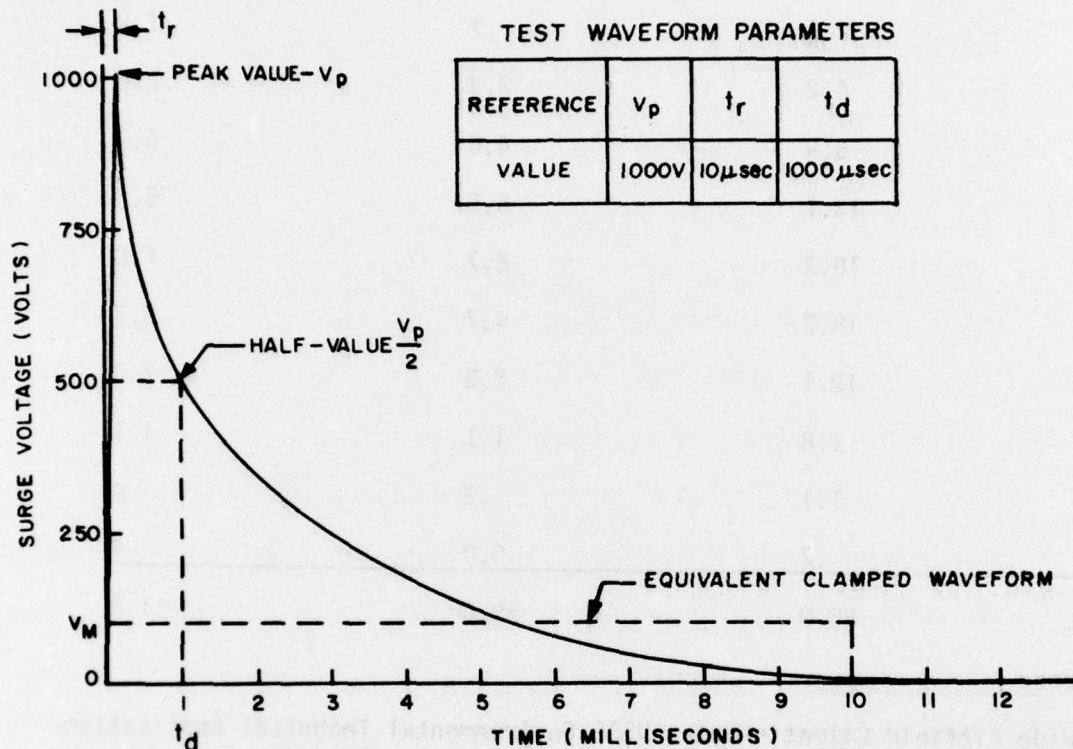


Figure 3 Test Waveform for Lightning-induced Voltages on Buried Control Cables

### III. AN ANALYSIS OF THE SSILS:

A conclusion was reached, early in the analysis, that total elimination of damage potential to the SSILS would not be practical. The installation criteria for a landing system (see Figure 4) are such that subsystems such as the localizer and glideslope stations attract lightning. Most of the inter-connecting electrical cables around the airport supporting the ILS are buried telephone lines. While these buried cables are normally protected against

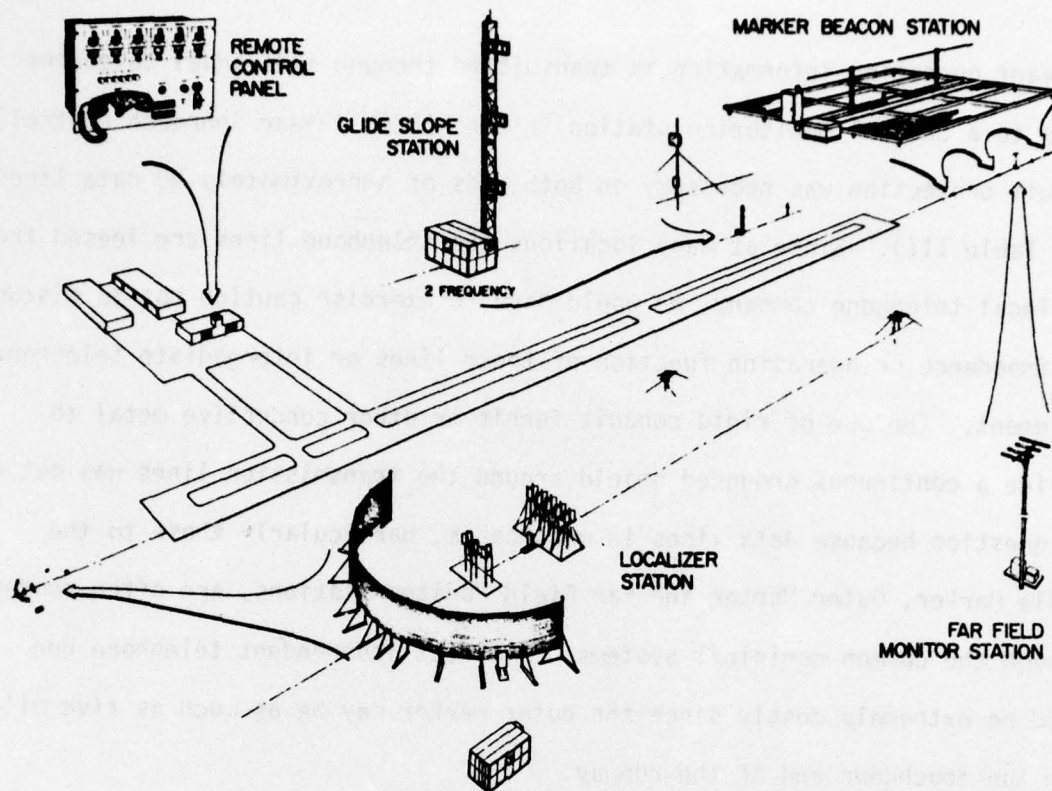


Figure 4 Instrument Landing System AN/GRN-27(v)

direct lightning strokes, they are frequently subjected to transients caused by strokes which hit the earth's surface nearby. The current from these strokes in seeking the path of least resistance to true ground may travel long distances along a communications cable. Since total shielding is not available, the pulses are coupled to the data lines and carried directly into the electronic circuitry of the equipment. When the voltage level of the pulse exceeds the withstand value of the interface components, catastrophic failures occur. A design goal to achieve 95% protection was established for this program.

The AN/GRN-27 SSILS has sensitive monitoring circuitry at each of the radiating subsystems. The performance status of these monitors and other



relevant operating information is transmitted through individual telephone lines to a central monitoring station in the RAPCON (Radar Approach Control). Circuit protection was necessary on both ends of approximately 50 data lines (see Table III). Since at many locations the telephone lines are leased from the local telephone company, we would have to exercise caution not to disrupt the impedance or operating function of these lines or intermediate telephone equipment. The use of rigid conduit ferris or other conductive metal to provide a continuous grounded shield around the transmission lines was out of the question because data lines in most cases, particularly those to the Middle Marker, Outer Marker and Far Field Monitor Stations, are often routed through the common municipal systems. A direct independent telephone run would be extremely costly since the outer marker may be as much as five miles from the touchdown end of the runway.

#### IV. AVAILABLE PROTECTION DEVICES

It was determined that protection would be made at the equipment level as close as possible to the utilities/ILS transition point. Data line protection values were determined and appear in Table III. Some of the available protection devices evaluated are:

A. Spark Gaps are composed of two metal electrodes separated by a dielectric. The gap may be sealed in a glass or ceramic container and the dielectric may be air or inert gas. Judicious selection of gases provides different ionization levels as a voltage is applied. Spark gaps are very reliable low cost devices. They have a very low voltage drop during the conduction state and have a high current carrying capability. Operation is bilateral. Among the disadvantages are the requirement for a relatively high spark over voltage and the requirement that the arc must be extinguished by

removing the voltage.

R. Diodes are available under various names depicting specific application. Some examples are the Zener Diode used to clip a transient or voltage and hold it to a rated voltage level: Forward Conducting Diodes include the standard silicon and germanium types used to bypass undesired signals: Controlled Avalanche Rectifiers are silicon diode rectifiers with controlled reverse characteristics: "TransZorb," a trade name used by General Semiconductor Industries, Inc., are zener diodes having an extremely fast response time. Diodes are generally small and have a wide assortment of characteristics which make them suitable for line/circuit interface application. They may be selected to monitor critical voltage levels with extremely fast (2.0 nanoseconds) response. Diodes have the disadvantage of not being able to tolerate large voltage surges for long periods of time.

C. Metal Oxide Varistors (MOV's), a product of the General Electric Company, are similar in characteristics to bilateral coupled zener diodes. Their response time is in the order of 1.0 to 3.0  $\mu$ seconds, dependent on the voltage and rise time of the pulse. The current carrying capability of the MOV is better than that of a diode, but not as good as the spark gap.

D. Silicon Controlled Rectifiers (SCR's) can be used as a unipolar spark gap to suppress transients. They have excellent surge-current ratings but they must be triggered by an external source. The SCR is a relatively expensive device.

E. Hybrid Devices are available in a variety of sizes and ratings, all prepackaged. These are normally multistage devices using diodes in one stage to reduce the steep wave front of the transient by clamping it at a given level; a second stage may contain a high current dissipating device such as a

TABLE III

## WIRE INTERCONNECTION VIA TRANSIENT SUPPRESSOR BOX REMOTE MONITOR STATION

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
D1TB1		E1TB-2	OUTER MARKER	
7	Control A	1	0-12	
8	Control B	2	0-12	
5	Abnormal	3	0-28	
1	OFF	4	0-28	
	Standby	5		Not Used
3	Main	6	0-28	
12	28 VDC	7	28 $\pm$ 5	
4	28 VDC	8	28 $\pm$ 5	
D1TB1		E1TB-3	Middle Marker	
7	Control A	1	0-12	
8	Control B	2	0-12	
5	Abnormal	3	0-28	
1	OFF	4	0-28	
	Standby	5		Not Used
3	Main	6	0-28	
12	28 VDC	7	28 $\pm$ 5	
4	28 VDC	8	28 $\pm$ 5	
D1TB1		E1TB4	Inner Marker	
7	Control A	1	0-12	
8	Control B	2	0-12	
5	Abnormal	3	0-28	



TABLE III Continued

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
1	OFF	4	0-28	
	Standby	5		Not Used
3	Main	6	0-28	
12	28 VDC	7	28 $\pm$ 5	
4	28 VDC	8	28 $\pm$ 5	
B2A10TB1		E1TB5		G1ideslope
15	Control A	1	0-12	
16	Control B	2	0-12	
12	Abnormal	3	0-28	
14	OFF	4	0-28	
10	Standby	5	0-28	
9	Main	6	0-28	
11	28 VDC	7	28 $\pm$ 5	
13	28 VDC	8	28 $\pm$ 5	
7	Intercom	9	-55 - +55	
8	Intercom	10	-55 - +55	
A2A10TB1		E1TB6		Localizer
15	Control A	1	0-12	
16	Control B	2	0-12	
12	Abnormal	3	0-28	
14	OFF	4	0-28	
10	Standby	5	0-28	
9	Main	6	0-28	
11	28 VDC	7	28 $\pm$ 5	

TABLE III (Continued)

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
13	28 VDC	8	28 $\pm$ 5	
7	Intercom	9	-55 - +55	
8	Intercom	10	-55 - +55	

## WIRE INTERCONNECTION VIA TRANSIENT SUPPRESSOR BOX MARKER BEACON STATION

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
E1TB2-3-4		D1TB1		Outer-Middle- Inner
4	OFF	1	0-28	
6	Main	3	0-28	
8	28 VDC	4	28 $\pm$ 5	
3	Abnormal	5	0-28	
1	Control A	7	0-12	
2	Control B	8	0-12	
7	28 VDC	12	28 $\pm$ 5	

## WIRE INTERCONNECTION VIA TRANSIENT SUPPRESSOR BOX GLIDESLOPE STATION

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
E1TB5		B2A10TB1		Remote Monitor Relay K1
6	Main	9	0-28	
5	Standby	10	0-28	Relay K2
7	28 VDC	11	28 $\pm$ 5	
3	Abnormal	12	0-28	Relay K4
8	28 VDC	13	28 $\pm$ 5	

TABLE III (Continued)

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
4	OFF	14	0-28	
1	Control	15	0-12	
2	Control	16	0-12	
9	Intercom	7	-55 - +55	
10	Intercom	8	-55 - +55	
MISALN DEY		1TB1		Equip Cabinet
E	MISALN	6	12	
F	MISALN	7	0-12	
NEARFIELD MON		1A35P1		
4A3TB1	28 VDC	1	28±5	

## WIRE INTERCONNECTION VIA TRANSIENT SUPPRESSOR BOX LOCALIZER STATION

Wire From	Signal Name	Wire to	Operating Voltage (Volts)	Remarks
C1A16TB3		A2A10TB1		
8	DDM 2	2	+5.11	Far Field Mon
9	DDM Common	3	Floating	Far Field Mon
7	DDM 1	1	+5.11	Far Field Mon
5	Contact Common	6	28	
1	Contact A	4	0-36	
3	Contact B	5	0-36	
E1TB6				
9	Intercom	7	-55 -55	Remote Monitor
10	Intercom	8	-55 -55	Remote Monitor



TABLE III (Continued)

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
6	Main	9	0-28	Relay K1
5	Standby	10	0-28	Relay K2
3	Abnormal	12	0-28	Relay K4
7	+28 Volts	11	0-28	
8	+28 Volts	13	0-28	Relay K3
4	OFF	14	0-28	Relay K3
1	Control	15	0-12	
2	Control	16	0-12	

## WIRING INTERCONNECTION VIA TRANSIENT SUPPRESSOR BOX FAR FIELD MONITOR

Wire From	Signal Name	Wire To	Operating Voltage (Volts)	Remarks
A2A10TB1		C1A15TB3		
4	Contact A	1	0-36	
5	Contact B	3	0-36	
6	Common	5	28 $\pm$ 5	
1	DDM 1	7	$\pm$ 5.11	
2	DDM 2	8	$\pm$ 5.11	
3	DDM COM	9	Floating	

spark gap.

#### V. LIGHTNING TRANSIENT PROTECTION FOR THE SSILS:

Since it was desired to fabricate, install, test, and demonstrate a transient suppression modification on an active system, actual circuit changes were kept to a minimum. Transient suppression circuits were fabricated on separate boards for "in line" installation between the utilities transition point and SSILS interface circuitry. Circuit design was based on suppressing a pulse having the characteristics presented in Figure 3 to a safe level. The protection would be required to handle not only a simple pulse but a burst of as many as forty pulses at a 100 Hz rate. Power line protection would be at the 2000-volt peak value since power is generally provided over long distances via overhead wires.

##### A. Data Line Protection

The basic circuit initially decided on was a hybrid, illustrated in Figure 5. Signalite CG-75 and CG-110 spark gaps were selected as the primary arrestor devices. Since ionization of gas within these devices occurred in the order of approximately 64 and 85 volts respectively, additional circuitry was needed since these exceeded the rating of SSILS interface devices. General Semiconductor Industries series IN5555 (Military Specification MIL-S-19500/434, Reference 7) TransZorbs were selected as the secondary arrestor. The IN5555 is a zener which has an extremely fast reaction time, in the order of 2-3 nanoseconds; therefore, the incoming transient is immediately clamped to a safe level for the equipment. Resistor R1 in Figure 5 provides a slight time delay and the impedance needed to permit voltage to continue rising across the spark gap after the zener clamps. Several microseconds after the transient pulse appears, the gas tube ionizes, providing a low impedance path to ground and

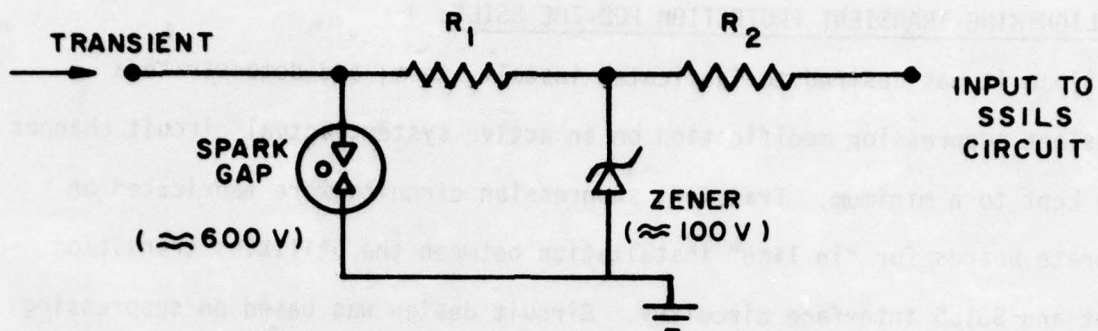


Figure 5 Basic Arrestor Circuit

protects the zener (transorb) from burnout. This current time relationship is illustrated in Figure 6. The R2 serves as a current limiter and impedance matching device to prevent short circuiting the interface components to ground while the spark gap is conducting.

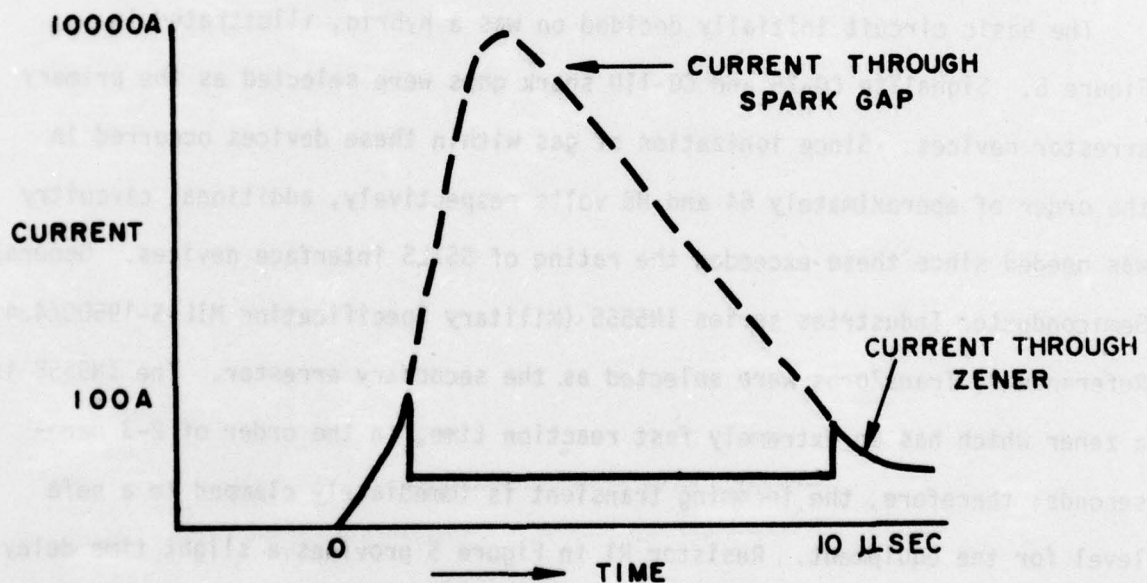


Figure 6 Current Through Protector Circuit

Each of the signal lines was individually analyzed, then parts selection was made to keep the number of different parts to a minimum and still provide adequate protection. It was determined that this protection could be provided



at two-line levels, 32 volts and 74 volts. A number of the data lines contained AC signals, thus needing bipolar devices. The Signalite CG-75 and CG-110 spark gaps met the requirement for bipolar operation for, respectively, the 32 and 74 volt lines. Transorbs L-39A and L-100A are back-to-back zeners of the IN5555 series encapsulated as a single part. The same basic circuit was repeated for each line, tailored as required. An overall module fabricated for each of the principal components of the SSILS is illustrated in Figures 7 through 11.

After initial testing, the basic circuit (Figure 5) was modified in two areas (Figure 12.) Resistor R3 was added as a current limiting device to enhance the longevity of the spark gap. Manufacturers' data sheets indicate a relatively short life for the spark gap, in the order of 40 strokes. Although tests indicated this figure to be conservative, the life expectancy was increased by a factor of at least 10 with the addition of the limiting resistor. A second significant change was made to the grounding buss. The ground side of the spark gap is immediately returned to the grounding stake at the utility pedestal, keeping the ground loop as short as possible to shunt the high transient potential away from sensitive circuitry. Other components are connected through chassis (system) ground to true ground providing the necessary isolation. This is illustrated in the schematic diagrams in Figures 13 and 14.

Physically, the protection circuits were wired onto printed circuit boards and placed in metal junction boxes. Photographs of the wired circuits for the Marker Beacon and Glideslope stations are shown in Figures 15 and 16. In each case the input cable was dressed into the bottom of the box and connected to the respective protection circuit along the left side. The ground sides of the



3

2

1

TRANSIENT PROTECTOR  
BOX

REVISIONS				
ZONE	LTR	DESCRIPTION	DATE	APPROVED
	A	REVISED WITH EXTENSIVE CHANGES.		
	B	VR1-VR3 CHANGED TO L-39A.		

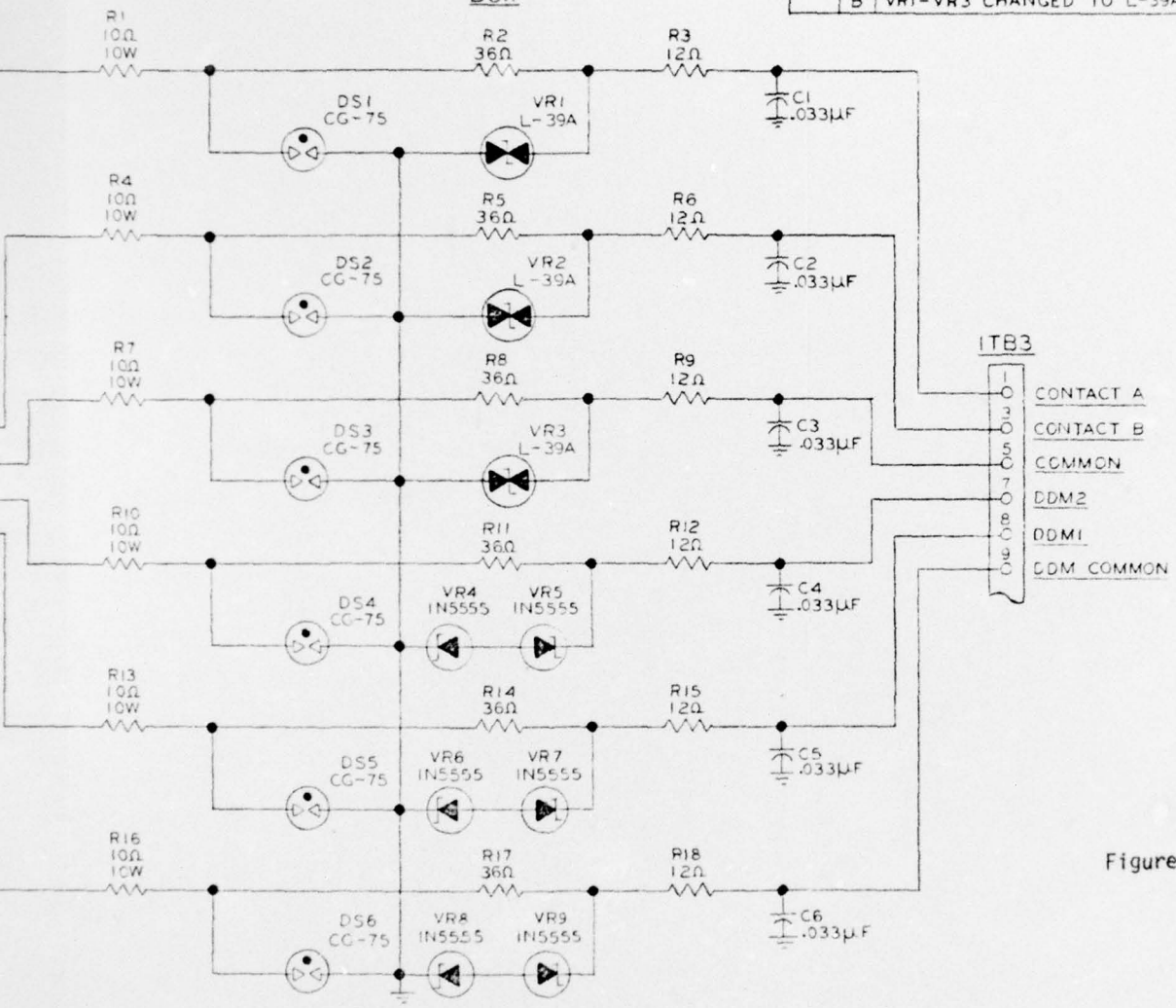


Figure 7

TRANSIENT SUPPRESSOR MODIFICATION

ED.  
LY TO INPUT SIDE OF CIRCUIT.  
CIRCUIT & BOND TO SYSTEM GROUND.

QTY	SYM	NUMERICAL	CODE	IDENTIFYING NO.	SPECIFICATION	MATERIAL	UNIT	ZONE	FIND
REQD			IDENT				WT		NO
LIST OF MATERIALS									
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES					U.S. AIR FORCE				
FRACTIONS TOLERANCES ON DECIMALS ANGLES					RADCC				
+ .XX ± .XXX ± °					TITLE				
MATERIAL					SCHEMATIC DIAGRAM- MONITOR, RADIO FREQUENCY, MC-9026/GRN-27(V)				
DRAFTSMAN					SIZE				
CHECKER					A.F. CODE IDENT NO				
ENGINEER					DRAWING NUMBER				
A.F. PROJ ENGR					C 07877 x7518629				
CONTRACT NUMBER					SCALE				
A.F. DESIGN ACTIVITY AUTHENTICATION					SHEET				
NEXT ASST					APPLICATION				
NEXT ASST					QTY REQD				

ENGINEERING DRAWING LAYOUT C

U.S. GOVERNMENT PRINTING OFFICE: 1963 O-780-078

2



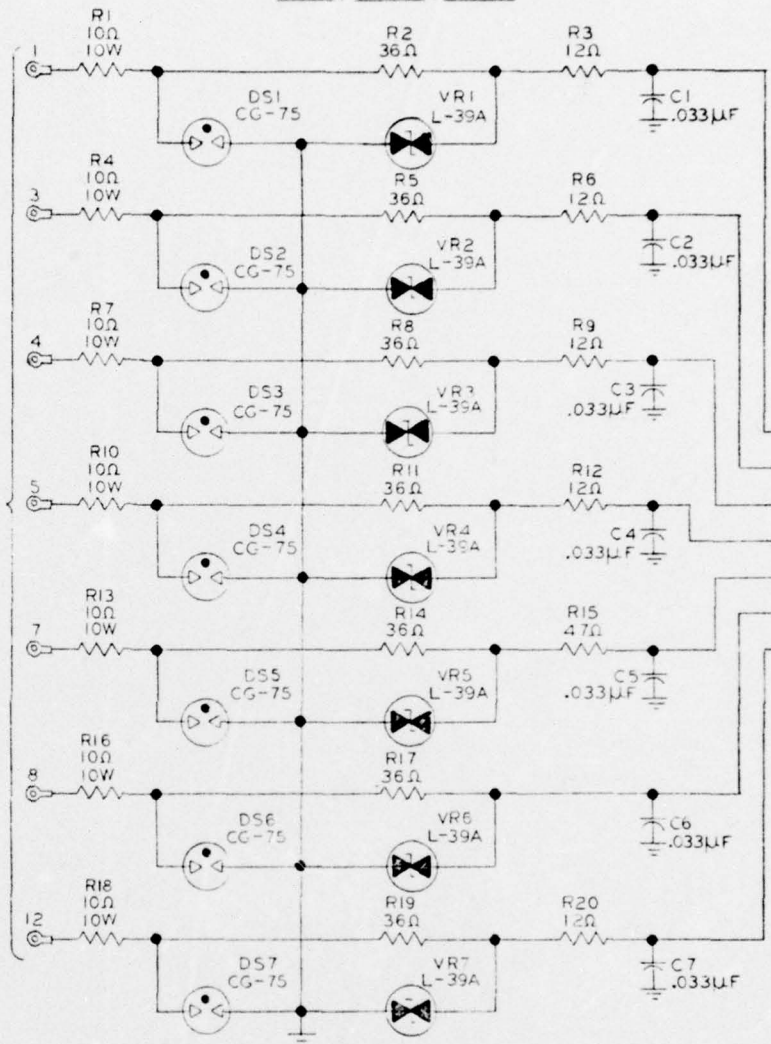
4

3

2

# MARKER BEACON STATIONS (INNER, MIDDLE, OUTER)

TO REMOTE  
CONTROL MONITOR  
CONNECT TO  
EITB2, TB4 OR TB3



## NOTES:

- 1- ALL RESISTORS ARE 1W EXCEPT AS NOTED.
- 2- DRESS GROUND FROM SPARK GAPS SEPARATELY TO INPUT SIDE OF CIRCUIT.  
DRESS ALL OTHERS ALONG OUTPUT SIDE OF CIRCUIT & BOND TO GROUND.
- 3- CG-75 IS SIGNALITE DESIGNATION FOR TRANSIENT PROTECTOR DEVICE.
- 4- ZENER DIODES CLAMP AT 33V TO 47V.
- 5- 10W RESISTORS SHOULD BE NONINDUCTIVE TYPE.

## TRANSIENT SUPPRESSOR M

QTY REQD	SYM	NOMENCLATURE	CODE IDENT
		UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES	DRAFTSMAN
		FRACTIONS TOLERANCES ON DECIMALS ANGLES	CHECKER
		± .XX ± .XXX ±	ENGINEER
		MATERIAL	A.F. PROJ. ENGR.
			CONTRACT NUMBER
			A.F. DESIGN ACTIVITY AUTH.

APPLICATION	QTY REQD
7518 GCF	
NEXT ASSY	USED ON
NEXT ASSY (FINAL AND)	

3

2

# MARKER BEACON STATIONS (INNER, MIDDLE, OUTER)

REVISION			DATE	APPROVED
ZONE	LTR	DESCRIPTION		
	A	REVISED WITH EXTENSIVE CHANGES.		
	B	VR1-VR7 CHANGED TO L-39A & NOS. ON DITBI CHANGED. *	2/2-11	

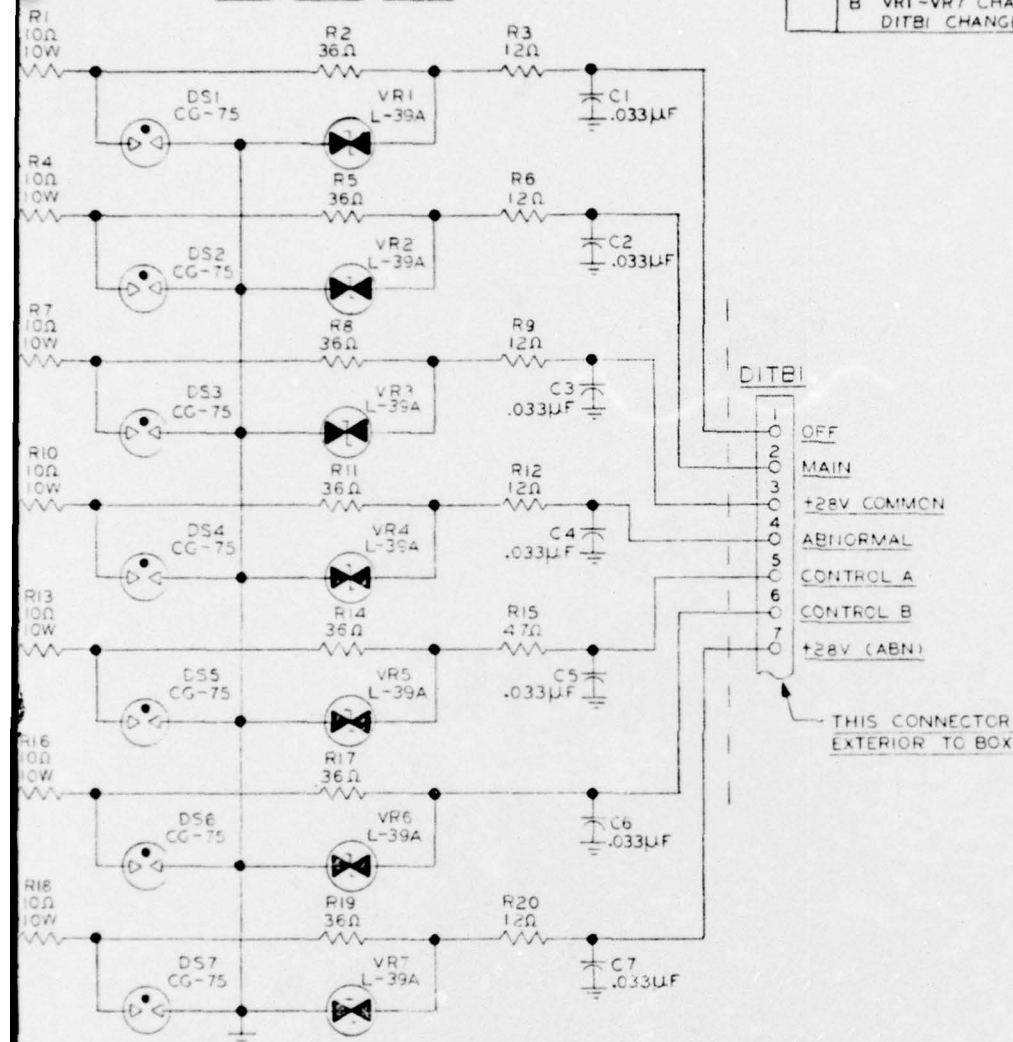


Figure 8

## TRANSIENT SUPPRESSOR MODIFICATION

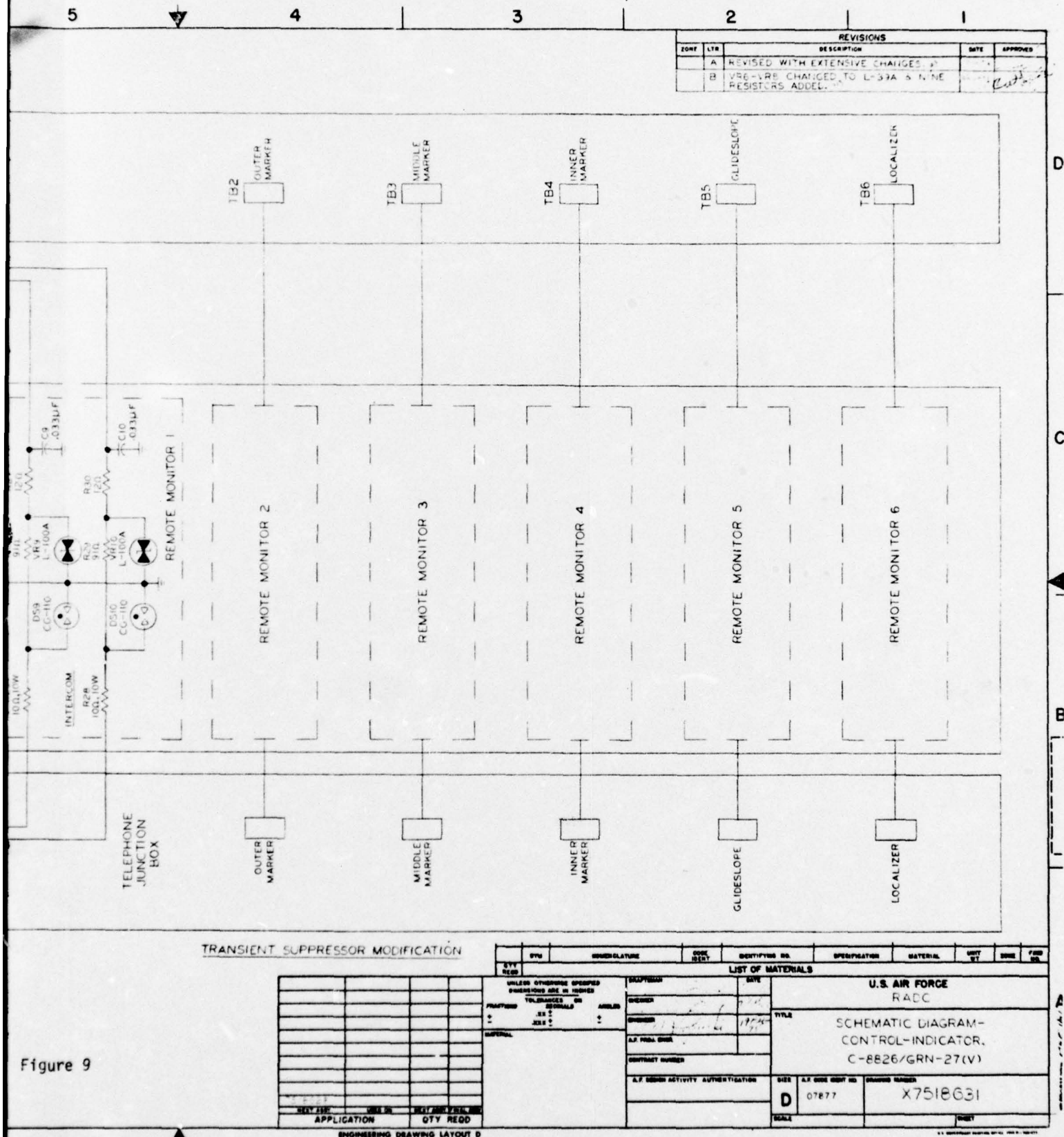
INPUT SIDE OF CIRCUIT.  
IT & BOND TO GROUND.

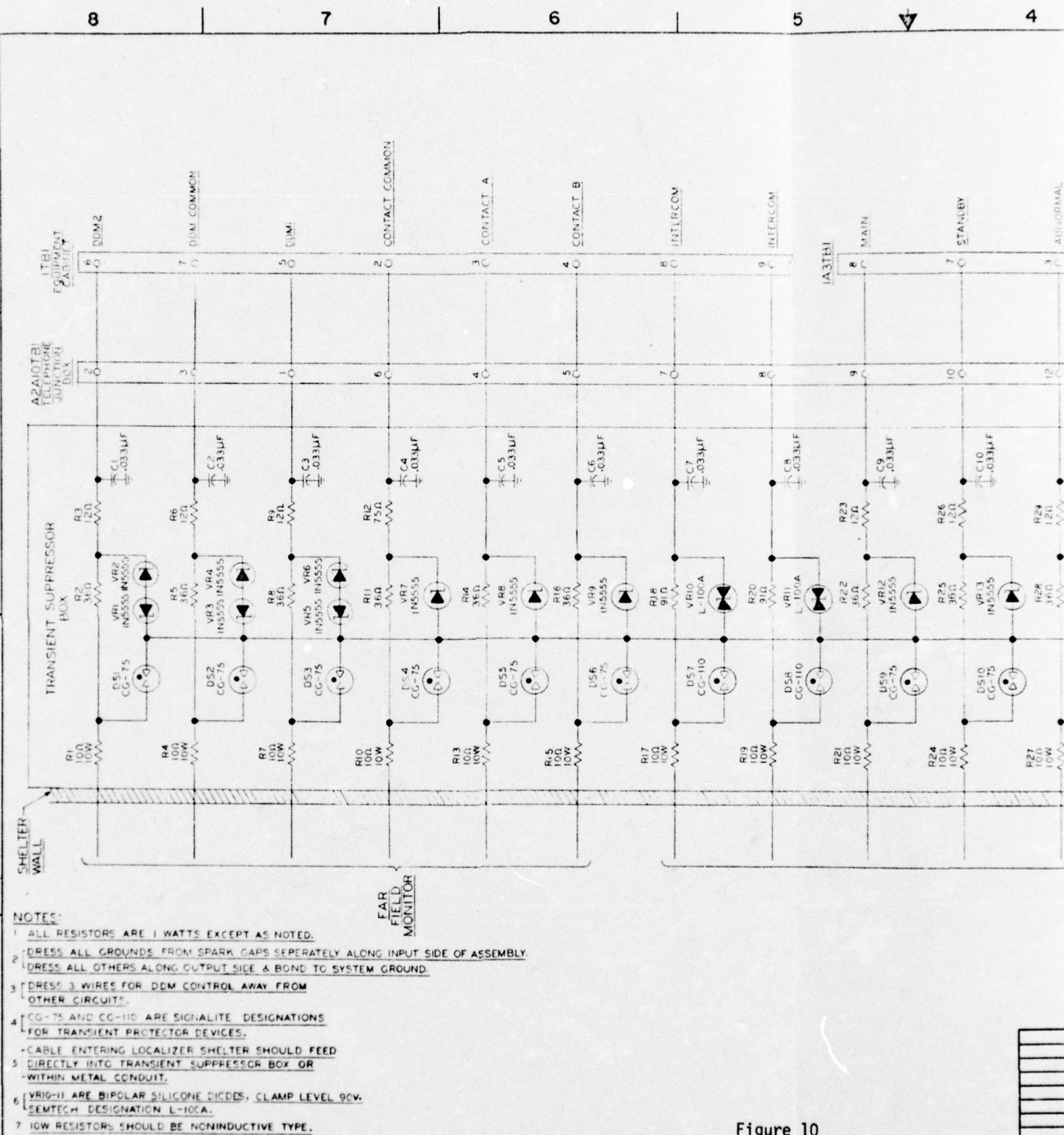
QTY	SYM	NOMENCLATURE	CODE	IDENTIFYING NO	SPECIFICATION	MATERIAL	UNIT	ZONE	FIND
REQD			IDENT				WT		NO
LIST OF MATERIALS									
DRAFTSMAN CHECKER ENGINEER A.F. PROJ. ENGR CONTRACT NUMBER A.F. DESIGN ACTIVITY AUTHENTICATION					U.S. AIR FORCE RADC TITLE SCHEMATIC DIAGRAM - BEACON SET, RADIO, AN/GRN-26 SIZE C 07877 DRAWING NUMBER X7518630 19				
UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES FRACTIONS TOLERANCES ON DECIMALS ANGLES .XX ± .XXX ± MATERIAL					SCALE SHEET				
APPLICATION QTY REQD					ENGINEERING DRAWING LAYOUT C				

2



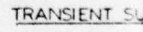












### TRANSIENT SURGE

- ### TRANSIENT SURGE

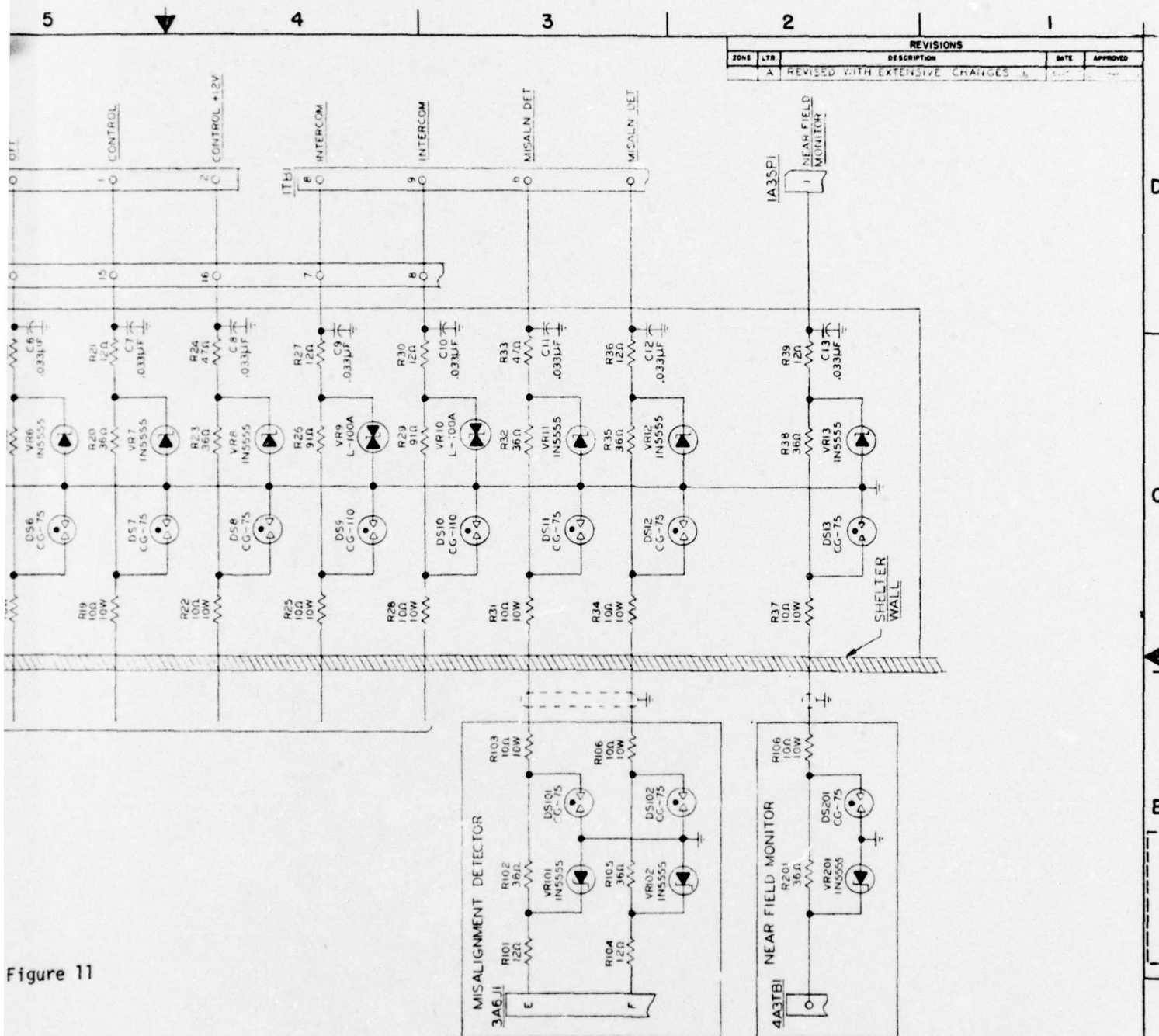


Figure 11

TRANSIENT SUPPRESSOR MODIFICATION

REV	DATE	DESCRIPTION	BY	CHKD	DATE	REVISION NO.	DESCRIPTION	MATERIAL	UNIT	QTY	REV	DATE
1												

LIST OF MATERIALS			
U.S. AIR FORCE			
RADC			
TITLE			
SCHEMATIC DIAGRAM -			
GLIDESLOPE STATION			
AN/GRN-27(V)			
A.F. DESIGN ACTIVITY AUTHORIZATION		DATE	DATE
D		07877	X7518633

REV	DATE	DESCRIPTION	BY	CHKD	DATE	REVISION NO.	DESCRIPTION	MATERIAL	UNIT	QTY	REV	DATE
1												

ENGINEERING DRAWING LAYOUT D

2

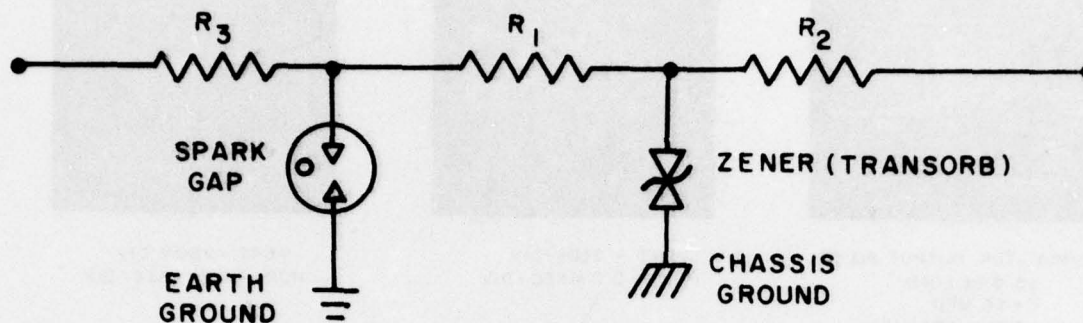
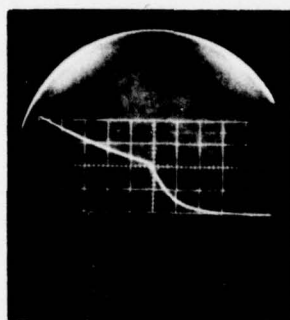


Figure 12, Modified Arrestor Circuit

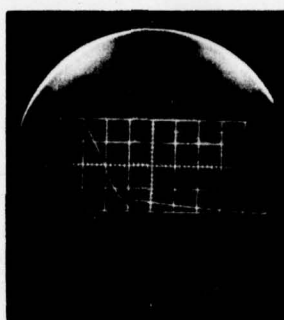
spark gaps were connected to a large ground buss which was returned to the grounding stake at the utilities' pedestal. Filtered signals appeared on the right side of the circuit boards. These were routed, via shielded wires, away from the input cables, to the appropriate circuits of the SSILS.

The circuits discussed were the principal line of protection placed directly onto the signal lines of the SSILS. Various other modifications were incorporated into the existing circuitry of the instrument landing system to enhance protection or the possibility of false indications which may be attributed to transients. These included diode shunts, hold-off circuits and time delays. The little independent circuit board appearing at the bottom of Figure 16, for example contains a time delay circuit used in resetting the glideslope monitors from the remote monitor position.





SIMULATOR OUTPUT PULSE  
25 OHM LOAD  
C = 56 MFD  
VERT - 250 V/DIV  
HOR - 0.2 m SEC/DIV



VERT - 250V/DIV  
HOR - 0.2 mSEC/DIV



VERT - 250V/DIV  
HOR - 0.2 m SEC/DIV

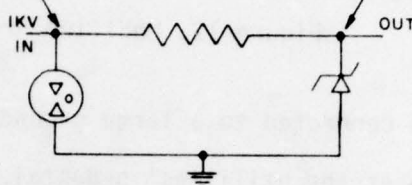
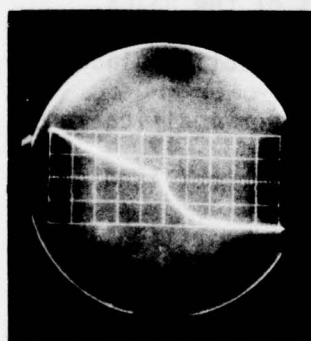
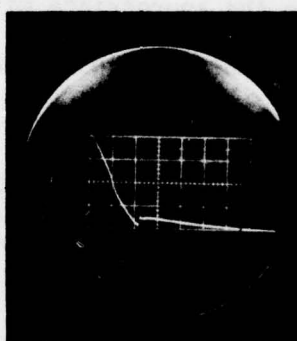


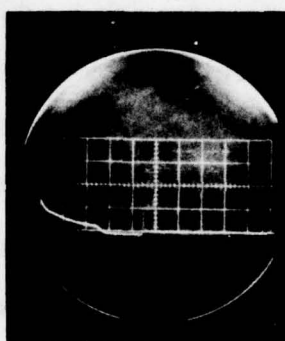
Figure 13 Transient Effects Using Common Ground



SIMULATOR OUTPUT PULSE  
25 OHM LOAD  
C = 56 UFD  
VERT - 250 V/DIV  
HOR - 0.2 m SEC/DIV



VERT - 100V/DIV  
HOR - .2 mSEC/DIV



VERT - 100V/DIV  
HOR - .2 mSEC/DIV

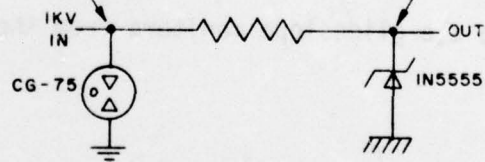


Figure 14 Transient Effects Using Separate Grounds

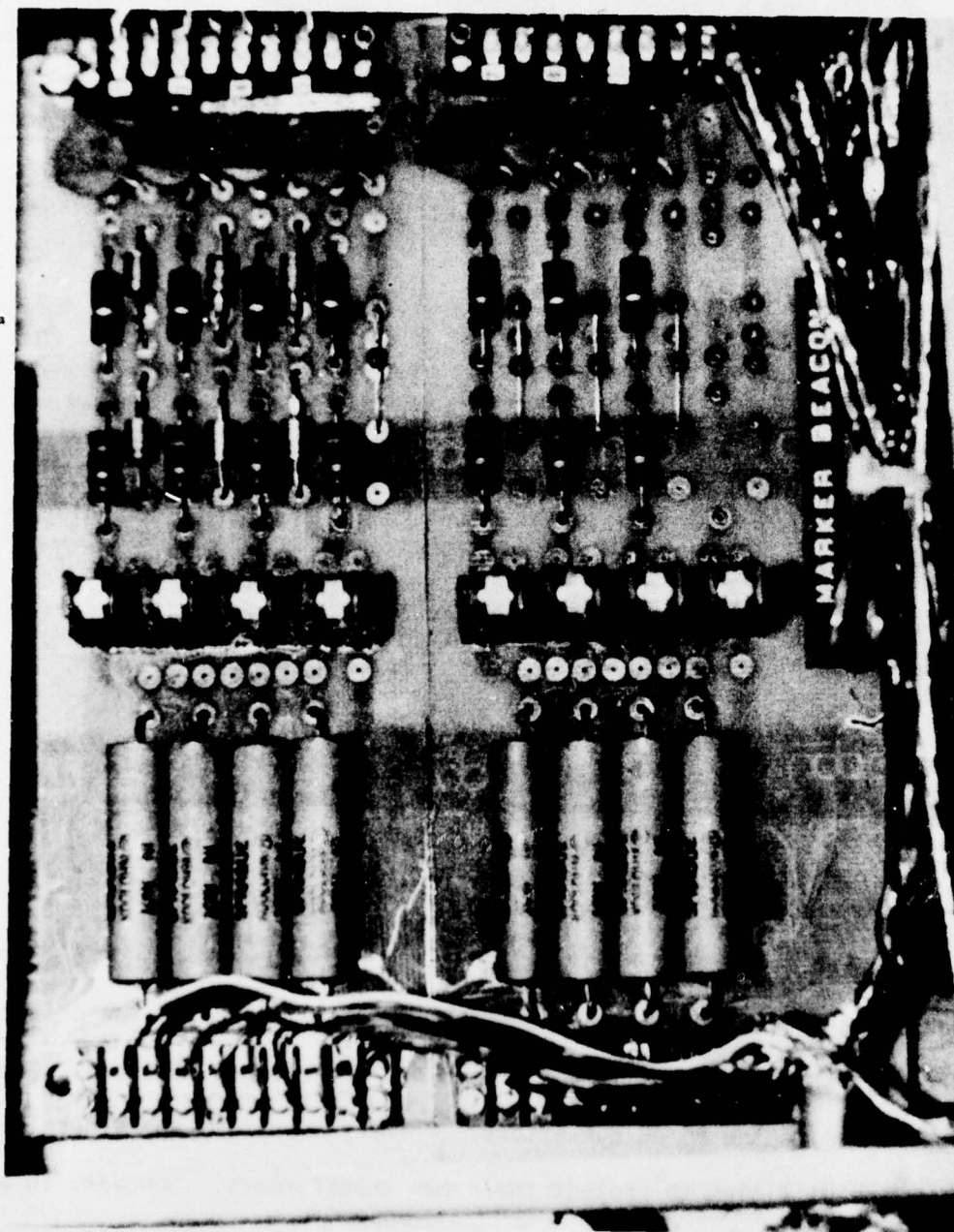


Figure 15 Wired Circuit, Transient Protection for Marker Beacon

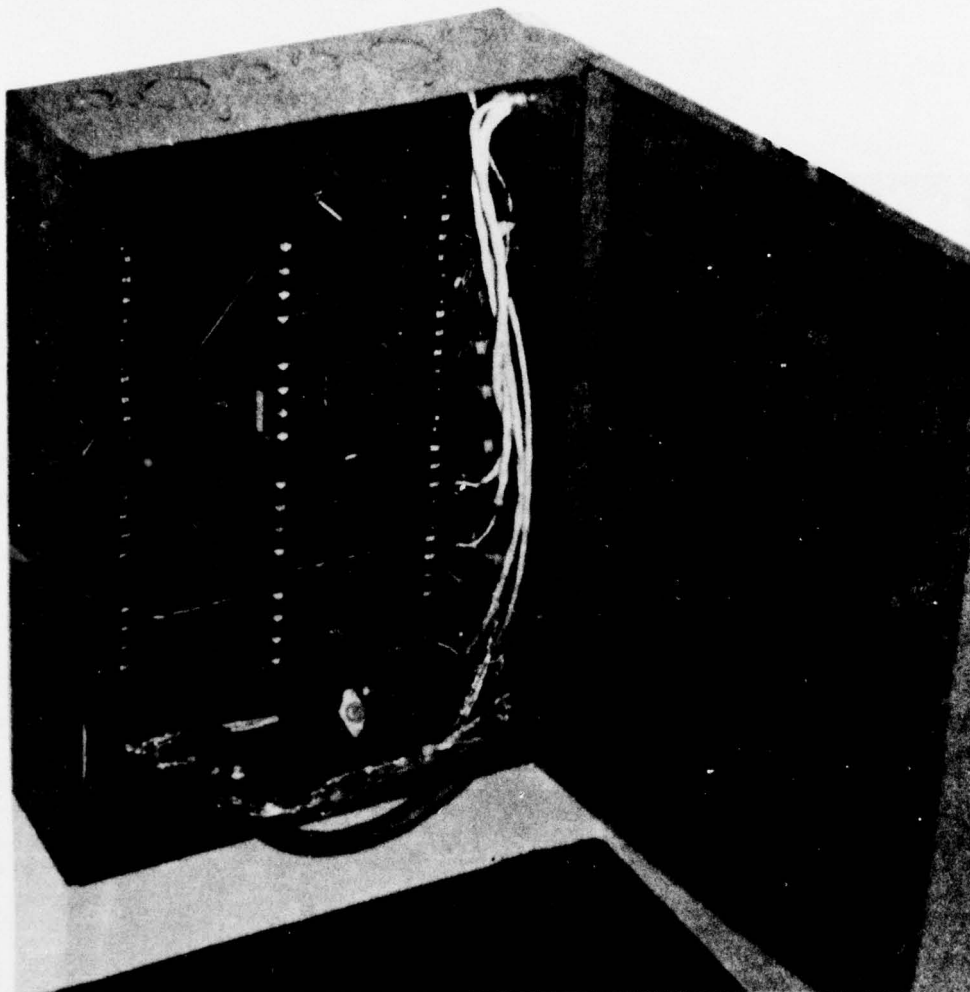


Figure 16 Wired Circuit, Transient  
Protection for Glideslope

B. Power Line Protection

Although the input power lines are technically isolated from the critical SSILS circuits, through the battery bank and battery charger, failures were being experienced in the DC/DC convertors. Primary arrestors are generally provided by the utilities to protect their own transformers. However, in an ILS installation this may be at a substantial distance from the equipment shelter. It was considered advisable to place secondary arrestors at the localizer, glideslope, and marker beacon stations. The far field monitor,







co-located with the middle marker beacon, was protected by a common unit. The arrestor modification used in the localizer and glideslope stations is shown in schematic form in Figure 17. The Joslyn Type #1235-01 was selected for these stations. Joslyn Protector Type #1230-01 was used at the marker beacon stations. Since these are sealed hybrid units, neon lights were employed on each power buss to indicate an operational protector. When the protector shorts to ground, which is the normal failure mode, the light is extinguished, indicating the failure. A photograph of the installed circuit at the Glideslope Station is shown in Figure 18.

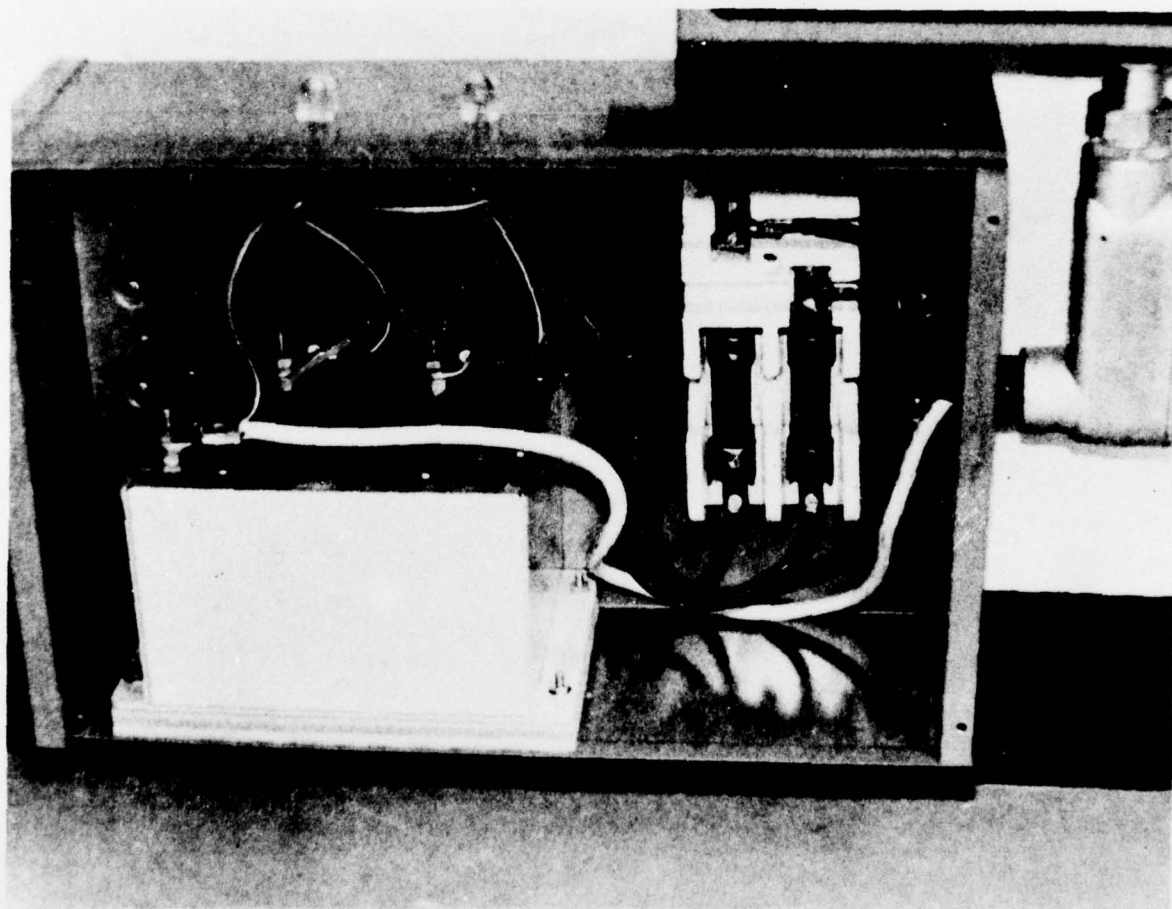


Figure 18 Prime Power Line Protection Circuit,  
Glideslope and Localizer Stations



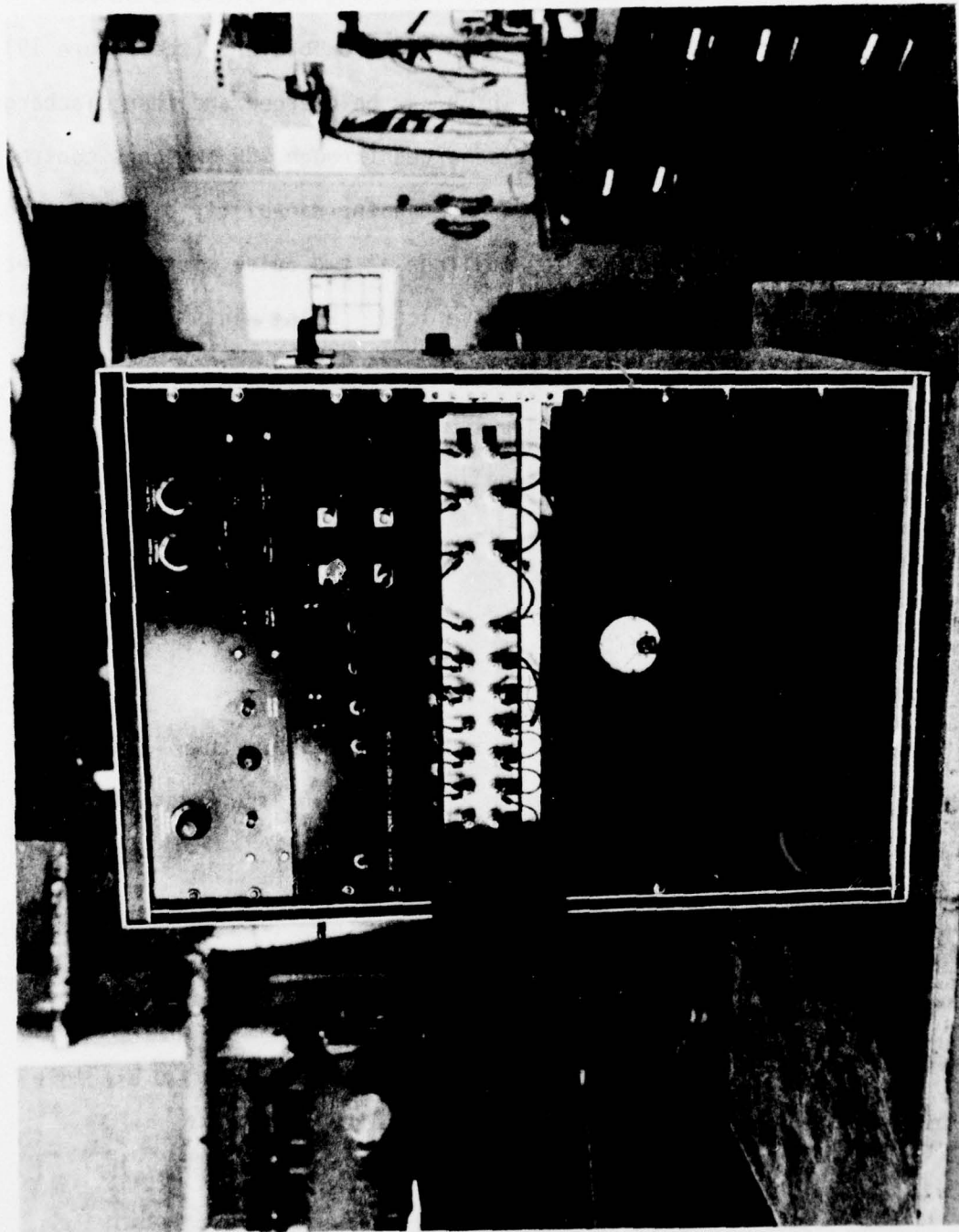


Figure 19 Lightning Simulator

## VI. LIGHTNING SIMULATOR:

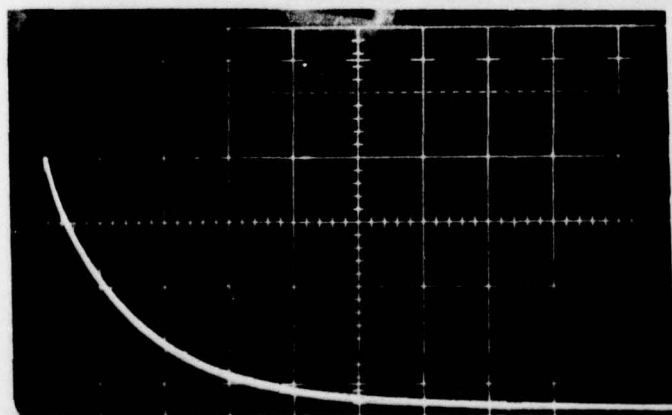
A simulator capable of reproducing the lightning transient pulse was designed and fabricated within the RADC High Power Laboratory (see Figure 19). It contains a bank of pulse capacitors which can be charged and then discharged on demand. The selected wave form is discharged through SCR switches controlled by associated logic circuits. The simulator has the capability of generating 1 to 100 pulses, 100 to 1000 volts in amplitude at two pulse widths, 100  $\mu$ sec and 1 msec across a nominal 25 ohm resistive load. Pulse width and energy are determined by capacitor selection, i.e.

1. 100 msec pulse 8 MFD 1KV = 4j/pulse

2. 1 msec pulse 58 MFD 1KV = 29j/pulse

$$\begin{aligned} E &= 1/2 CV^2 \\ &= 1/2 (58 \text{ MFD}) (10^3)^2 \\ &= 1/2 (58) \\ &= 29 \text{ joules/pulse} \end{aligned}$$

A typical photograph of the output pulse is shown in Figure 20. Additional details on the simulator are published in RADC-TM-76-10 "The Solid State Lightning Simulator" by Joseph J. Polniaszek.



Vertical - 200V/Div  
Horizontal - 1.0 msec/Div

Figure 20 Output Pulse From Lightning Simulator (25 ohm load)

## VII. FIELD TESTING:

Field testing was conducted in two phases: Phase I, a simulated test on a "hot-mock-up" and Phase II, implementation of an active operational AN/GRN-27(v) system. Details of these tests are documented in the "Final Test Report, AN/GRN-27(v) SSILS Lightning Fix Project" published by Electronics Systems Division (ESD), which was prepared by the AFSC/AFCS Test Team. In each phase of testing, design personnel implemented the protection modification and conducted a preliminary test to verify that the modification functioned properly and system operation was not being degraded. A Test Team consisting of AFSC-ESD and AFCS personnel then conducted an intense acceptance test.

The basic test setup is shown in Figure 21. Both ends of each data line under test were monitored while the transient was superimposed on the line. This test pulse was inserted at the connecting point for SSILS equipment and the telephone line at both ends of each line that was tested. Test pulses were started at 100 volts and graduated in 100-volt increments to peak voltage of 1000 volts. Each line was pulsed a minimum of six times with a single pulse followed by a burst of 40 pulses. Figure 22 shows the advancement of the pulse in amplitude. The reproducibility of the waveform is particularly worthy of notice since it is relatively constant even in the burst mode.

It was during the Phase I testing period that circuit refinements were derived for direct application to the AN/GRN-27. It was also during this period that the change was made to the basic circuit from that shown in Figure 5 to that of Figure 12. Sufficient confidence was gained that the protection circuits were adequate to protect the SSILS circuitry from harmful transients reaching an amplitude of 1000 volts peak. With this confidence, the refurbished, refined protection circuits were installed on the operational



AN/GRN-27(v) at McGuire AFB, New Jersey. The same sequence of tests was again performed first by the designers, then the impartial AFSC/AFCS Test Team. After three months of testing by the Test Team and exposure to the natural environment, the modification was deemed acceptable to the User.

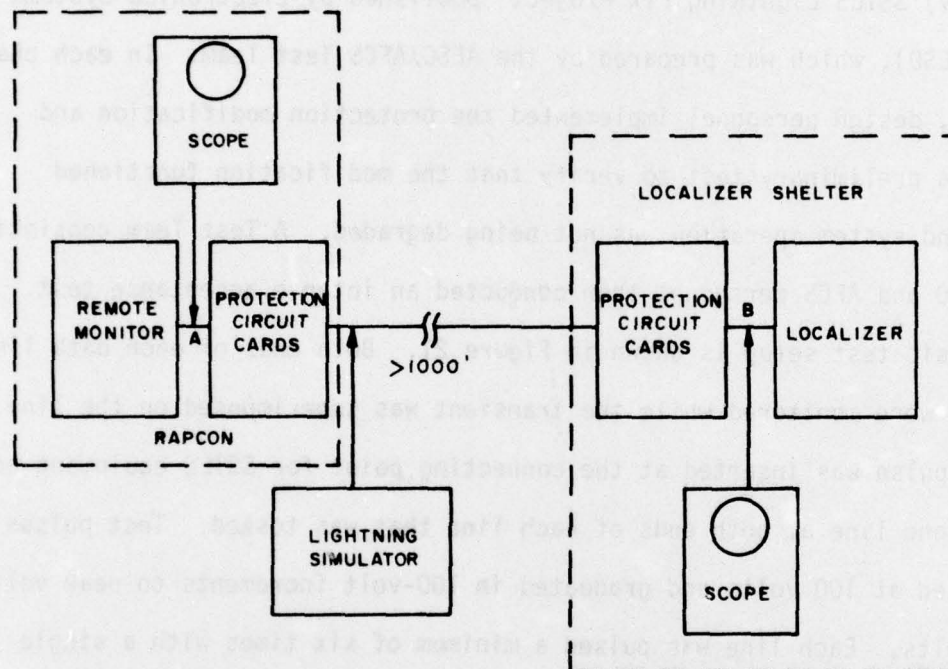
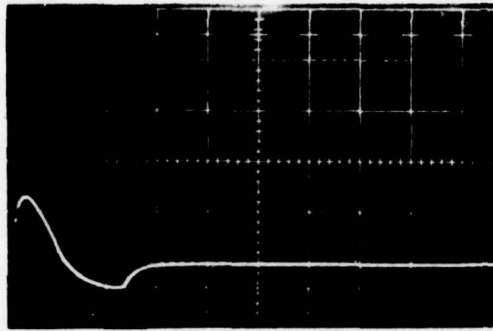


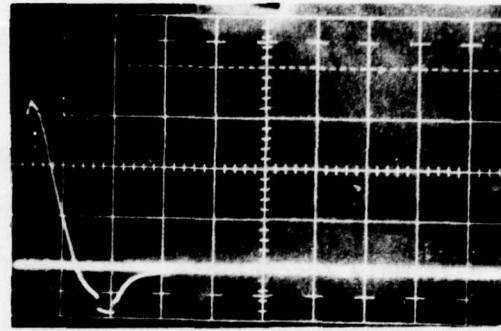
Figure 21 Typical Test Set-up At Operational Facility - Lightning Transient Test

#### VIII. CONCLUSIONS AND RECOMMENDATIONS:

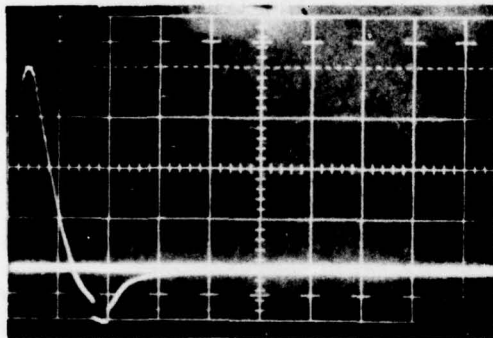
It was found that documentation of data and experiences relating to the characteristics of transients generated by lightning is not readily available. Since detailed data was lacking and time and funding were not adequate for an in-depth monitoring and data collection study, a number of assumptions were made relative to the transient characteristics. It may be derived that at least some of these assumptions were made correctly since the AN/GRN-27 at McGuire AFB is presently sustaining its second season of successful operation with the transient protection modification. Lightning is one of the oldest



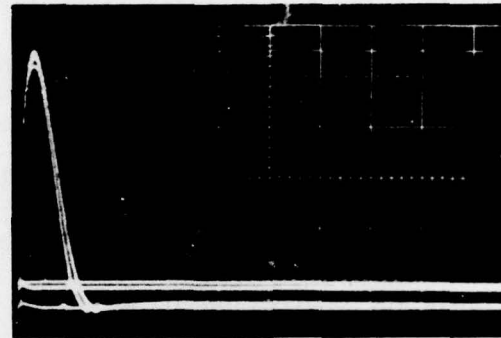
200 v



700 v



1000 v



40 Pulse Burst

Figure 22 Test Pulse Sequence Lightning Transient Test  
Vertical - 20v/Div  
Horizontal - 0.1 milsec/Div

phenomena known to man. We have some relative knowledge of the potential energy it may contain, but know very little about the characteristics and side effects of this energy. Whereas tube type devices are relatively nonresponsive to transients, transistors and integrated circuits by nature of their design are extremely susceptible. This could be catastrophic, particularly when these devices compose the input/output circuits of equipment like the instrument landing system. The many antennas and interconnecting cables are virtual collectors and transmission lines for transients seeking true ground.

Transient protection for equipment which may be exposed to such an environment should become a design requirement for Air Force and DOD equipment. Judicious selection of parts will help to protect the inner circuit which may have susceptibility even with interface protection. A number of good devices suitable for protection are becoming available and may be applied directly to the operational circuits.

It is recommended that a program be initiated by the Air Force to:

- A. Study the characteristics of lightning.
- B. Develop a good set of characteristics applicable for various types of systems.
- C. Establish a standard, such as MIL-STD-461 and MIL-STD-462, outlining the withstand requirements and proven, acceptable methods of testing for compliance.
- D. Evaluate and recommend approval or limitations of protection devices.
- E. Evaluate and certify test equipment.



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5. Polniaszek, Joseph J., Jr., "The Solid State Lightning Simulator" Rome Air Development Center RADC-TM-76-10, June 1976
6. Final Test Report, AN/GRN-27(v) SSILS Lightning Fix Project, Electronics Systems Division, Sept. 1975
7. Military Specification MIL-S-19500/434 27 February 1970, Semiconductor Device, Diode, Silicon, Transient Suppressor TX and Non-TX Types IN5555 through IN5558 and IN5610 through IN5613

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